

2. FLT 79 WAS A STABILITY INVESTIGATION AND SUCCESSFUL

DEMONSTRATIONS OF ADEQUATE ELEVON EFFECTIVENESS DOWN TO

129 KEAS (125 KEAS ON AIRCRAFT 123), AT 30,000 FEET WERE MADE.

ELEVEN POINTS WERE TAKEN FROM 295 TO 129 KEAS. MAXIMUM THROTTLE

SETTING 93 PERCENT, MINIMUM THROTTLE 86 PERCENT, MAXIMUM RATE

OF SINK 1000 FPM. AIRCRAFT WAS SHOWN TO BE FULLY CAPABLE

OF RECOVERY AS LOW AS THE 125 KEAS MENTIONED ABOVE BY USE OF

ELEVON ONLY I.E. NO POWER CHANGES.

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- FLT 80 WAS MADE TO CHECK VELOCITY IN A SHALLOW DIVE AT CRUISE POWER 97 PERCENT AT 34,000 FEET. AIRCRAFT WAS PUSHED OVER TO ZERO 'G' AND 10 PERCENT DIVE WITH NO POWER CHANGE AND DESCENDED TO 25.000 FEET AT WHICH POINT A SPEED OF 377 KEAS AND .92M WAS REACHED WITH NO FURTHER INCREASE IN SPEED POSSIBLE UNDER THIS CONDITION.
- THE SECOND PART OF FLIGHT 80 WAS TO DETERMINE THE BLEED OFF IN SPEED AT 97 PERCENT IN A SHALLOW CLIMB FROM 34,000 TO 38.000 FEET AT 2000 FPM. AIRCRAFT REACHED 192 KEAS AT 38.000 LEVELED OFF AND THEN CONTINUED TO SLOW DOWN UNTIL 180 KEAS WAS REACHED. AT THIS POINT IT BECAME EVIDENT THAT THIS ALTITUDE COULD NOT BE HELD WITHOUT AN INCREASE IN POWER AND SPEED WAS BLEEDING OFF AT 1 KNOT PER 6 SEC.
- THE THIRD PART OF FLT 80 WAS MADE TO CHECK A SIMULATED TURN AS DONE BY AT 34.000 FEET. .84 M. 96 PERCENT RPM. AND 12,000 POUNDS OF FUEL THE AIRCRAFT WAS PUT INTO A 30 DEGREE LEFT BANK FOR 110 DEGREES HOLDING THE SAME ALTITUDE. AT THIS POINT THE THROTTLES WERE RETARDED THREE AND ONE-HALF PERCENT RPM, THE 30 DEGREE BANK MAINTAINED AND A 2 DEGREE CLIMB SET UP. AFTER ANOTHER 110 DEGREES OF TURN REACHING AN ALTITUDE OF 35.600 FEET. THE SPEED HAD BLED OFF TO 195 KEAS AND .6 M.
- NO FURTHER TESTS WERE RUN 4 JUNE 63. THE RESULTS ARE BEING STUDIED AT THIS TIME.

END OF MSG

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Approved For Release 2000/05/05: CIA-RDP71B00590R000200030011-0

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